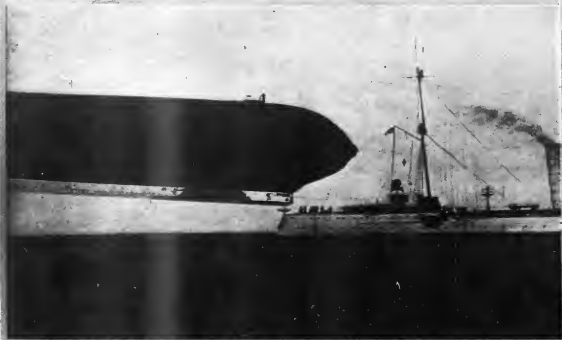


FEBRUARY 14, 1921

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AVIATION AND AIRCRAFT JOURNAL



German Airship Mooring to a Cruiser Steaming at Full Speed

VOLUME X
Number 7

SPECIAL FEATURES

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JAPAN?

INTERNALLY TRUSSED WINGS
AIR SERVICE PROMOTIONS

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FEBRUARY 14, 1921

AVIATION AND AIRCRAFT JOURNAL

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The Internally Trussed Wing

By William B. Stout

Early airplanes were built largely from the nonstressable materials of wood. From a study of the air loads on curved or shaped surfaces, it was found that the early style of analysis continued clear up to the wing and planes were designed around wing curves of the best "half-elliptic ratio" with small variations in the height of the thick sections, some with and some without camber. If a wing curve had a half-elliptic ratio of 36 as against a camber of 12, it was found that the curve with the highest half-elliptic ratio was the curve, provided the latter was within proper range for possible loading speeds.

When the idea of internal trussing of wings was first presented, reports of the so-called experts of the day were against the idea, because of the weight of the thick sections, and they were not sufficient in their half-elliptic as related to other wing curves, although the advantage was admitted of the thick wing sections at slow loading speeds.

In a study of thick wing sections, or any wing sections, three things are fundamental: first, the L/D ratio, second, the depth of spars for structural ligaments, and third, the center of pressure movement, and the latter has been neglected as a function of lightness and control in an airplane, but in thick wing sections, it becomes vital, the reason which will be explained.

The ordinary thick wing section, such as the tailpipe type used by the Germans, has the center of pressure movement at the leading edge of the wing. The L/D ratio, 1/3 the total chord of the wing. This means that the total surface for proper controllability must be mounted on a long lever arm formed by the fuselage, and at a distance to the rear of the wing of about three times the chord. This structure must be strong, and therefore, weighty.

In any wing curve at speed, three-fourths of the total lift is gained by the upper surfaces, while the lower surfaces give but 25 per cent of the lift. The lift should be kept in mind preliminary to a proper explanation.

In the German ordinary wing curve, the bottom surface is concave or partially so, and as thick wing sections, curves at the nose and concave toward the trailing edge, giving the advantage of "tailpipe" sections given to the German wing.

In adopting a thick wing, we give up curves known to have half-elliptic ratio, or any curved wing, and we adopt curves with half-elliptic ratio of 12. This is done because it was found in our first internally-trussed ship completed in America that a plane could be built of this 12 curve much lighter than a ship with any curved wing, and it was proved that an aerodynamic efficiency, or lift in the wing, was more than given back by the fact that the wing did not have to lift so much per square foot.

In the first step, the difference between the American and German method of attack in the internally-trussed problem is that of adopting a "long chord wing" section, instead of the so-called "thick wing" section. Suppose structure required for a wing curve of 36 or any curve being adopted, would a proper factor of safety. The German method would be to make a thick wing section of, say, five times the depth, or a 60 in. chord at the root of the wing where it joins the fuselage and taper to 20 in. at the tip. This "tailpipe" type, to obtain depth of wing sections, a long chord would be adopted, and for the 18 in. depth, the wing chord at the root would be, say, 32 times the wing depth, or an L/D chord of the wing would be 1/32 the fuselage. In other words, the proper structure required for strength is obtained, although in the Root type, there is space for more spar. In the German type, one has the aerodynamic loading that the wing curve is not good at high speed on account of thickness, like the steady nose lead. In the long chord type, however, the wing still keeps an efficiency up to very high speeds, so that a greater range of aerodynamic is possible. The German type is not good at all related advantages in this structure in that more area can be

built in the same weight in the long chord type, so that as speed increases, it is a larger, lighter airplane.

In the above cases, for comparative reasons, the wing is taken from the root to the tip so that the wing curve at the tip is of better known ratio of thickness to chord than it is possible at the root of the wing, on account of the requirements of structures. The chord at the tip is about the same in other cases, but the Advisory Committee Report No. 75, recently made, states that a surprising result of some recent experiments with the tapered plan view wing shows as much as 20 per cent more lift at the same speed as that of the rectangular shape, so that what was lost in the wing curve is what is known as the L/D factor, or in lift coefficient, is easily obtained by the added lift advantage of the plan view. The advantage, however, is very small, and the same in the long chord design, than in the German type.

Recent analysis are establishing a rule that the aspect ratio of an airplane can no longer be figured as the ratio of the mean chord of the wing to its total span, but in the ratio of the span to the chord of the tip, so it is the wing tip line that makes the advantage or disadvantage of aspect ratio a plane form.

All these things are changing the minds of designers, and there have come to a point where aerodynamic results are not taken as first requirements, but structure and possible weight per square foot have the basis of acceptance in question of the final plans. The analysis, therefore, in comparing wings, those which will enable one to get more lift per square foot weight per square foot of wing and per horsepower of engine installation, taking it for granted that if this lighter wing is used, where within the range of aerodynamic advantage it will have many superior points over other flying structures of greater weight and lower horsepower per pound. Some at the same time the internal trussing eliminates at one stroke the weight of the structure, and the consequent weight of the plane for a given performance, even as an engine a beginning as the German Junkers, then the end of the entire analysis cannot help but be a very much advanced airplane.

Let these structures be used on a "thick" (thick wing) plane requires a careful study of center of pressure movement as the wing curves used. Ordinary wing curves are not advisable as long chord wings. To obtain a stable wing curve where the lift is at the leading edge, the wing curve must be designed—these sections rather than tailpipe sections were designed—these sections having the same curve at the bottom of the wing as at top.

As explained before, the top surface of the wing requires about 75 per cent of the total lift at speed, so that to take tip, one can hardly lose more than 25 per cent of the lift at the wing. In internal trussing, however, and tapered wings, the tip, and the strength of a center spar is in proportion to the weight of the wing, and the weight of the wing is that in doubling the depth of the spar, by the change in wing curve, we can cut the weight of the plane more than in half. In other words, if we take a wing curve of 36, and a wing curve, we would still have increased 50 per cent in lift, and, that the curve is still 25 per cent to the good.

There is an added advantage, however, in that a long tail structure is not needed in the center of pressure movement on the curve, for the fact that the controls can be located at the trailing edge of the wing, just as the rudder is a small part at the trailing edge of the wing and steers it equally well. This gives one a great deal of safety in the fuselage, and of landing gear height, or fuselage depth another very much in its favor. Thus, in the first step are shown possibility of a great advance in planes and wings, and as related to lift, aerodynamic and structural advantages, and the latter for the major betterment of control, stability and speed means.

At the present time, planes can travel at little over 100 miles per hour, and speed up to 130 or better.

Japan?

The following excerpt from an article in *The Aeroplane*, London, England, January 15, by C. G. Grey, the editor, gives an interesting view of Japan's air status by a particularly well-informed English writer.

Just as it is possible to suspect a person's abilities without having any personal liking for him or her, so one may admire a nation's enterprise and self-reliance without liking the people and the Government of the country. The Japanese, however, are the Japanese nation, which one regards as the great menace to the peace of the Pacific. We may expect that before many years are past the alleged horrors of the press will be regarded as much more reliable than the meaning of the word "Pacific."

It is commonly known that Japan has for many years been building up a navy and an army which have already placed her among the foremost of the world's fighting powers. Japan is not troubled by any serious doubt making the world safe for democracy. Her line has the Japanese government troubled itself, as ours has done, about making the world safe for hypocrisy. In the next open and above-board fashion Japan has adopted the doctrine of Armed Peace as the nation's best safeguard. For this one has the greatest admiration and respect.

The Aerodynamic Lesson

So far back as March, 1920, it was reported that the Japanese Military Mission in Europe had bought 300 airplanes, chiefly Sophs and Fokkers, in France. Personally they were bought from the French Government, and they were bought for the Japanese or Russian Air Force being active in Japanese hands.

Back about the same time, General Nagakura, the Chief of the Japanese Air Force, was in England, and he and the nation with the best air service will dominate its results both on land and sea. And subsequent events show that Japan is really up to this advanced position, which seems to be as strongly spoken by our publicists.

Recent Developments

It is common that Japan's fast big purchases of airplanes could have been made in France, seeing that England is only a few days' ride away. But it is not so much as is indicated by the fact that the Japanese army is in Japan and highly under French influence, while the Japanese navy is modeled on English lines. It is not so much as is indicated by the fact that the Japanese army is in Japan and highly under French influence, while the Japanese navy is modeled on English lines. It is not so much as is indicated by the fact that the Japanese army is in Japan and highly under French influence, while the Japanese navy is modeled on English lines.

As a result, however, the Japanese believe in an system of naval aviation. For they have engaged privately as instructors, both landward and on the flying side, certain very able officers, formerly of the R. N. A. S. and presumably intend to use them as staffs of the knowledge which they would have acquired from our staffs in England. Which they probably will do just as well and possibly at less expense.

A Technical Advantage

In this they will have something of an advantage over an official mission, for the officers and men of such a mission would only be permitted to fly machines which had been previously passed by the Committee on the Japanese Navy. As a result, it is to be seen, particular standard of safety. This standard (or factor, if you prefer it) is regarded by one of our best designers as being estimated by such highly scientific methods as to detect seriously from the performance

of our machines without in fact adding in the slightest degree to their safety.

That being so, free British pilots engaged by the Japanese navy will be at liberty to fly the best possible machines built in the defense of one's own possible defenses, first from official inspectors. And so it may come to pass that the Japanese flying services may actually be equipped with better aircraft, either they had or they had not, than our own navy, but it remained in equipping a foreign fighting service there is no need in looking about strenuous criticisms or the International Air Convention, and so each designer is free to do his best.

Who Is the Enemy?

Now, in the light of the very self-evident fact that Japan is undoubtedly equipping her already great army and navy with advanced flying services, it is necessary to consider how and where these services are likely to be used. Japan is not likely to be attacked by China, nor by Szechwan, Tibet, nor by the Rajah of Burmah (who seems to be the nearest independent monarch), nor by the Commonwealth of Australia. Who, then, is the enemy against whom these great contingents are being raised? There remains only the United States.

There is never any difficulty in finding a cause for war if two nations are faced to fight. A row between America and Japan, the most powerful of the world, is not likely to be avoided. America and Japan, and possibly in the British Isles, would suffer, even though both parties in either case were considered as law and justice by justice.

When and How?

The most interesting question is when the fight will come if it comes at all it must come soon. The great war U. S. Navy is not meant to fight us. And if Japan is to fight America, Japan must strike before that American navy is ready.

If Japan openly takes command of the Pacific, which the Japanese navy holds in fact today, she can then land troops in America at her leisure. And, be it remembered, Japan's Mainland is Japan, which is about the biggest island in the world, and her army is as strong as ours, and her navy is as strong as ours.

Japan would never be as foolish as to land troops on the Pacific shores of the United States, with the Hockley in front of her. Japan would never be as foolish as to land troops on the Pacific shores of the United States, with the Hockley in front of her. Japan would never be as foolish as to land troops on the Pacific shores of the United States, with the Hockley in front of her.

Meanwhile the Japanese fleet, with long-range guns and bomb-dropping airplanes flying from several carriers, would make life very unpleasant in Hawaii, San Francisco, and other cities near the flying lines. And before the attack the Japanese of the Pacific Coast would be quite considerable damage before striking (or before striking) and then (perhaps) American ships in the Hawaiian territory.

Preparations for Peace

Meanwhile it is up to the States to prepare for the attack which may come at any time after the next two years. The worst attack must come from the sea. Therefore the defense must be against the enemy's ships and bombing airplanes.

Against the former the only weapon is the long-range-dropping airplane. Half a dozen torpedo machines, carrying in all 100,000 lbs. of explosive, will account for any 15,000 lbs. building a long-range ship from the coast. Therefore a fleet of about a dozen torpedo machines stationed along the Pacific Coast will make that Coast safe against gunfire at a cost of less than one building.

But over and above them it is necessary to provide defense against the many's bombing machines which can leave their strips 100 miles or so from the coast. To meet these machines flying airplanes, anti-aircraft guns, and searchlights are needed in large numbers.

Furthermore, fighting machines capable of alighting on and getting off the sea will be needed to assault the torpedo machines in the water on the Japanese coast, for it may be assumed that the fleet will carry short-range high-performance fighters with which to beat off torpedo attacks. These machines are of a special kind to our airplane fighting machines, and here again we know nothing more than most people. It is to be hoped that in this direction the U. S. Army, or Navy (whichever has the job of defending the Pacific Coast) will not be too proud to learn from us.

Moreover, we can buy steel, iron, a fleet with submarines, Japan's army, and airplanes, and self-confidence. It would be at the bottom for this country if our political people had no such.

Tests for Aviator's Certificate

Instructions governing the method of tests for Aviator's Certificate as issued by the Aero Club of America under the rules of the International Aeronautic Federation are as follows:

Candidates must accomplish the following tests, each being a separate flight.

Test A—A flight without landing, during which the pilot shall remain for at least an hour at a minimum altitude of 2,000 meters above the point of departure. The descent shall finish with a glide, the engine cut off at 1,500 meters above the landing ground. The landing shall be made within 150 meters or less of a point B defined by the official markers of the test without starting the engine again.

Test B—A flight without landing around two points (or buoys) situated 500 meters apart, making a series of five figure-eight turns, each turn resulting out of the two points (or buoys). This flight shall be made at an altitude of not more than 500 meters above the ground (or water) without touching the ground (or water). The landing shall be effected by.

1. Finally shutting off the engine or engine at least when the aircraft touches the ground (or water).

2. Finally stopping the aircraft within a distance of 50 meters from a point fixed by the candidate before starting.

The candidate must be alone in the aircraft during the test.

In flights for hydroplaning certificate, starting from and landing on the water is permitted for all of the tests.

The course on which the candidate accomplishes the test must be marked out by two posts or buoys situated not less than 500 meters (547 yards) apart.

The turns round the posts or buoys must be made alternately to the right and to the left, so that the flight will consist of an unbroken series of five figure-eight turns.

The issuance of the certificate is always discretionary. The responsible representative of the Contest Committee is charged with the supervision of the tests and is responsible for the preparation and accuracy of the reports which must be dated, signed and returned to the chairman of the Contest Committee as soon as practicable after the completion of the test.

The responsible representative must see that the posts or buoys about which the aviators were to land at a distance not greater than 1500 ft (500 meters) apart, and shall take in the report the way in which the distance between the posts or buoys was measured, whether by chain, steel tape, or other means.

An observer must be stationed in the close vicinity of each post to see that the turns are made in the proper order, direction and number.

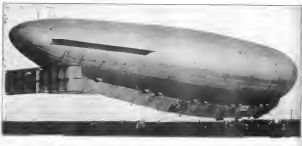
Under the heading of "Remarks" the time of day and wind, or conditions should be noted and the comment of the responsible representative as to the manner in which the applicant handled his machine.

The observer is required to fill in the answers to the following questions:

A. Duration and altitude. Did candidate remain at a minimum altitude of 2,000 meters throughout a period of at least one hour? Was the descent made from 1,500 meters with the motor cut off throughout? Was the alighting made in view of the observer? Was the alighting made in a satisfactory manner?

B. Distance flight. Was the flight made in accordance with provisions in Test B?—A. Alighting. Was the alighting made on land or water? Was the alighting carried out in a satisfactory manner? At what distance from the point did the aircraft come to rest? How was course measured and how marked?

The report must be signed by the representative of the Aero Club of America and also by that of the observer.



THE ITALIAN SEVERINO ANTONI ROMA, WHICH HAS BEEN PURCHASED BY THE U. S. A. AIR SERVICE

The Monocoque Fuselage

By Lester D. Seymour, E. E.

During the late war, among many other developments in the art of aircraft, there appeared a new plane called the wood fuselage for airplanes which, while appearing in this country, has not been exactly reproduced. The French word monocoque as applied to an airplane fuselage means simply a fuselage that is truly one piece and not built.

The present American approach is found in the Venetian motor built by the U. S. Air Service for the Gordon Bennett race. It is the purpose of this article to present for the interested, an building technique, the general method employed in this type of construction and to point out in not too great detail the advantages which it has been hoped to obtain.

After the war, while in Paris the writer was privileged to visit several factories building airplanes and among them one engaged in the manufacture of the true monocoque fuselage. Day the very rough notes made at that time and sketches of the better one available as a basis for this description, and consequently these can be given only approximate dimensions, etc.

The fuselage as constructed in the factory visited was built of thin laminations similar to those used in ordinary ply wood glued and bradded in place. The pieces used in building up this laminated ply wood were approximately 1/16 in. in thickness and from 3/4 to 7/8 in. in width, varying with

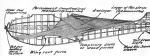


FIG. 1. FRENCH MONOCOQUE CONSTRUCTION

the position in which they were placed on the fuselage, representing 12 lb. per sq. ft. The laminations were described varied in length from 60 in. to more than 1 ft. but to pieces up to 15 ft. in length. The application of the laminations to the fuselage looking up the center or plywood was not as shown in Fig. 1. The procedure was:

A series of wooden discs or plates whose circumference fit the section of the fuselage at any particular point are placed in proper sequence and spaced at about 8 in. intervals on a steel tube about 4 in. in diameter, of a length just enough greater than the total length of the fuselage to enable the builder to place supports or "bones" under the ends in a position most advantageous for the workman. In applying the different layers of wood and adjust the longitudinal members. These disc forms are about 2 in. in thickness and held in their proper position on the tube or shaft before mentioned by means of wooden sleeves placed between each disc. The whole assembly is held in position by large bolts placed on either threaded end of the tube, holding all the discs in compression, each against the one next to it.

At the ends or tail end of the fuselage (the fuselage section is usually circular or slightly elliptical as built in the monocoque) a framework or internal frame forming the ribs of the horizontal stabilizer and vertical fin is set in position of the fuselage to be, these being usually disposed around the circumference of the circular cross of the form, each set in grooves or notches in the edge of the discs. At the forward end of the fuselage there is placed at this position where the

wing roots will be, wooden members attached to permanent rigidity of the fuselage at that point.

Thin longitudinal members or laminations must be cut and covered with impregnate as we know them, because they have an extra and are made in sections (on their member is usually) than an equal number of the most expensive wood is.

The application of the plywood is now begun. Beginning at the rear of the fuselage, the end of a strip of wood such as described above is placed to one of the longitudinal fuselage members and wound spirally around the fuselage toward the forward end, being fastened or latched at the proper places in the longitudinal members at an angle of approximately 45 deg.

This process is continued until the form is entirely covered with a series of these strips side by side spirally around fuselage and attached to the large of internal bracing of the stabilizer wing ribs and longitudinal bracing.

An opening is left in the fuselage for the pilot and observer to enter the cockpit, also an approximate place for the engine at the extreme nose.

After the wood is applied has been given one complete layer of lamination, the process is repeated, this time in a usual plywood practice, laying the strips of lamination in opposite direction and gluing the layers together. While constructing any particular layer, it is often best to place by small pieces



FIG. 2. FRENCH MONOCOQUE FUSelage COMPLETE

or blocks of wood tacked over the surface to maintain the necessary shape, pressure is then only placed at the position of the laminations or other bracing. This process is continued until the plywood has reached a thickness of five layers.

The fuselage will now appear (Fig. 2) to be wood moulded in the form of an airplane, nothing remaining except the actual wings. This conception is the most strong because the wood is being built in the shape as described, but confirmed exactly to all the curves of the wing ribs, stabilizer, vertical fin, and other and now remains with a ready appeal to be cut away, when represented in strengthened after the fuselage has been completely made and the first application of varnish applied. The rubber too, has been made of fused plywood so that in the finished piece the wings will be the only false part of the plane.

The frame or skeleton frame before mentioned are now removed by first sliding out the large tube and transferring the weight of the fuselage from the tube to "bones" now bearing directly upon the fuselage. The discs are subsequently and removed through either the cockpit or engine opening in the large end of the fuselage. The types which the writer saw completed were built both for rotary and Hispano-Born engines and the engine mounting on some of the fuselage was slightly different in each case to conform to the requirements by the two different engines.

At intervals necessary to give the required stiffness to the structure, light metal bulkheads were placed inside the fuselage. In the front end the engine supports were at-

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